

Space Development and Test Directorate

Department of Defense Space Test Program

"Access to Space"

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Missions:

(1) Develop, test, and evaluate National Security Space systems
 (2) Execute advanced space development and demonstration projects
 (3) Develop and launch small launch vehicles and missile defense targets
 (4) Rapidly transition capabilities to the warfighter

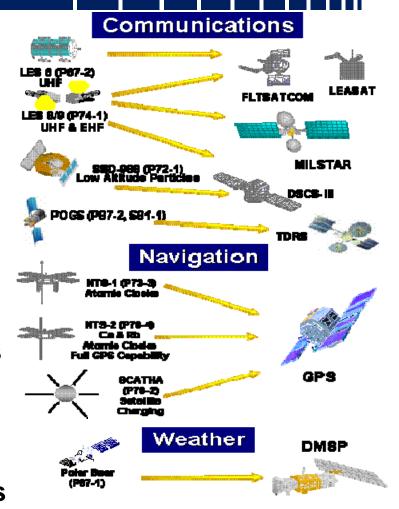


"One Team...Delivering Small, Responsive Space Capabilities"



DoD Space Test Program (STP)

- Chartered by OSD in 1965
 - First flight in 1967
- Primary provider of mission design, s/c acquisition, integration, launch, and on-orbit ops for DoD and DoD sponsored space experiments, technologies & demos
 - Ability to provide all spaceflight services <u>except</u> the experiment itself
- Single manager for all DoD payloads on the Space Shuttle and ISS
- Designated AFSPC's "Front door" for all auxiliary payloads seeking flight opportunities on DoD missions



STP has flown 493 experiments on Over 205 missions since 1967

(As of 10 June 10)

Access to STP Services

Space Experiments Review Board (SERB)

- Manifest based on:
 - SERB rank
 - Flight opportunities
 - Mission requirements
 - Available <u>STP</u> funds
 \$10M = SAF/USA approval
 \$10M = STP director approval

Reimbursable flight basis

- Manifest based on:
 - Flight opportunities
 - Available manpower
 - Available customer funds
 - SMC/CC and HQ AFSPC approval



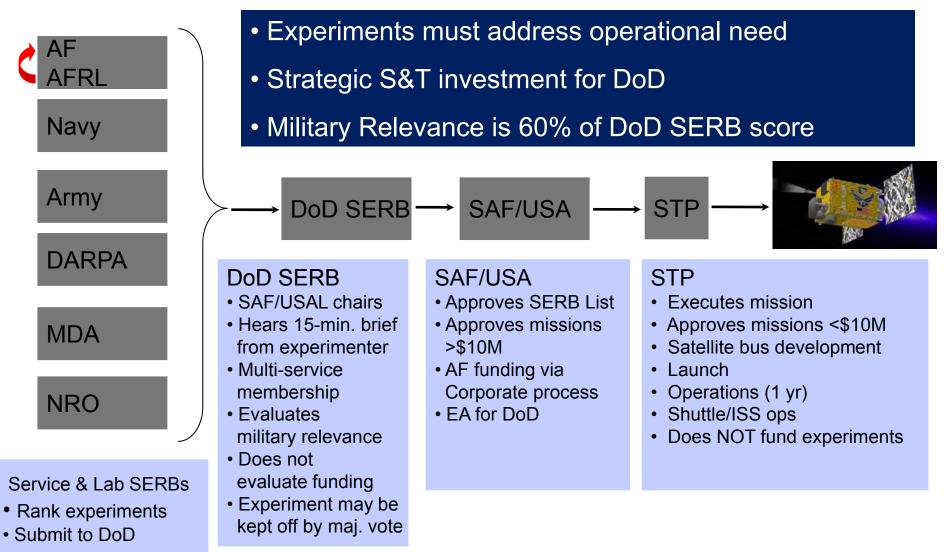


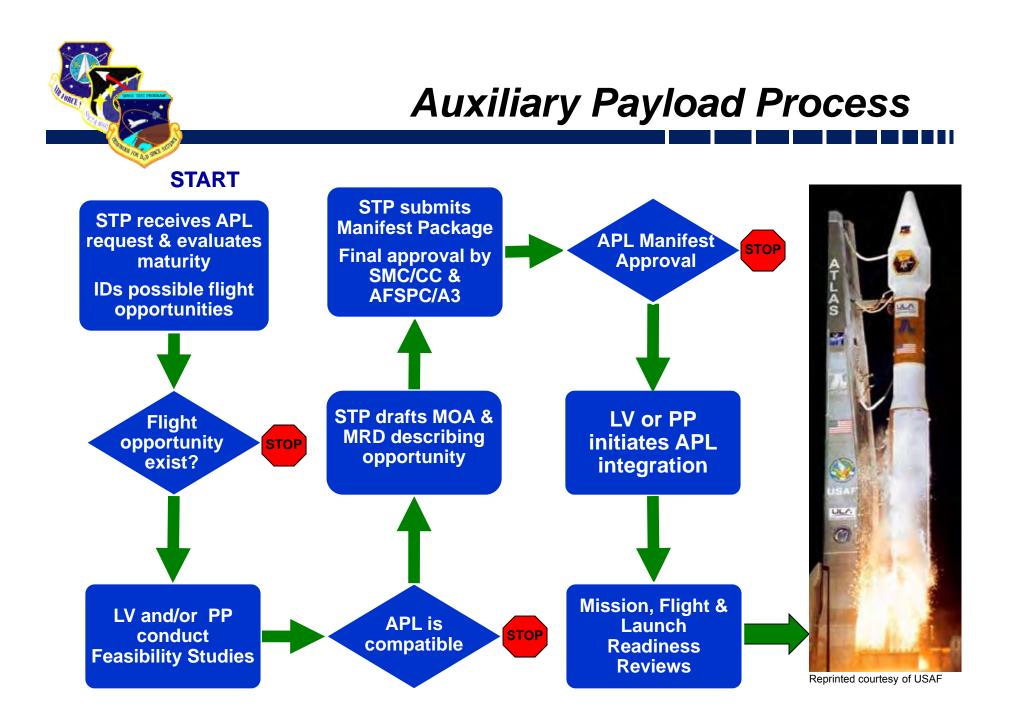






Space Experiments Review Board (SERB) Process







STP Access to Space

Spaceflight Methods:

- Shuttle/International Space Station (ISS)/Other NASA vehicles
 - Deployable, payload bay, mid-deck lockers, ISS internal/external
- Auxiliaries
 - Piggybacks payloads: leverage margin on existing SC
 - Secondary SC: leverage margin on existing LV
- Dedicated Launches (Minotaur, Falcon, Raptor, EELVs, suborbital sounding rockets)
- Also high-altitude balloons and zero-g flights



Atlas V

Zero G Flight



Setting Standards through Enablers

- Proactive measures to standardize SC design & construction
 - Based on 40+ years of lessons learned
 - Cohesive approach throughout space community
 - AFRL, ARMY, DARPA, NASA, NRL, NRO, SMC, along w/contractors and universities
- STP Enablers
 - EELV Secondary Payload Adapter (ESPA) CLASS SC
 - Standard Interface Vehicle (SIV)
 - Fast Affordable Science & Technology Satellite (FASTSAT)
 - Multi-Payload Adapter (MPA) Minotaur IV
 - Hydrazine Auxiliary Propulsion System (HAPS) Minotaur IV
 - Poly Picosatellite Orbital Deployer (P-PODs) / CubeSat
 - Accommodations on multiple LVs/SVs, Space Shuttle
 - Multi-Mission Space Operations Center (MMSOC) GSA

Maximize Launch Opportunities using Standardization



ESPA

(EELV Secondary Payload Adapter)

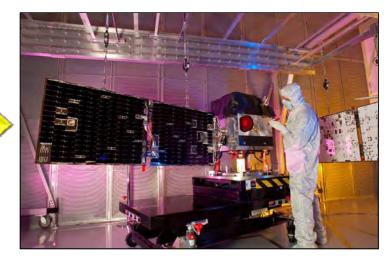
- SC weight ≤ 180kg
- SC Volume 35.5" x 28" x 24"
- CG Location < 20" from the SSIP
- Mechanical Interface 15"
- Electrical: Battery charge/monitor
- Fundamental Frequency > 35hz



SIV

(Standard Interface Vehicle)

- ESPA Class SC ≤ 180kg
- 1-4 Experiments
- Heritage subsystems
- Compatible w/ multiple LVs
- Compatible w/ MMSOC & AFSCN
- Storable until launch identified





FASTSAT: Fast Affordable Science & Technology SATellite (FASTSAT-HSV01)
Flight of up-to 6 SERB experiments
Sponsor: NASA MSFC
Size: 30" x 27" x 41" tall
Mass: 328 lbs



MPA

(Minotaur IV Multi-PL Adapter)

- Holds up to 4 ESPA Class SC
- Mass ≤ 180kg
- Volume 35.5" x 28" x 24"
- Maximizes lift capability
- First demo STP-S26 ILC May 10





Maximize launch opportunities using 180 kg satellites & secondary adapters



HAPS (Hydrazine Auxiliary Propulsion System)

- Dual orbits from small LVs
- Precise orbit insertion
- First demo STP-S26 ILC May 10



P-POD

(Poly-Picosat Orbital Deployer)

- 10x10x10 cm cube, ~ 1 kg mass ("1U")
- LVs: Rockot, Dnepr, Minotaur I, Falcon-1, Minotaur IV

1 U CubeSat (Cal Poly1)



Cal Poly P-POD



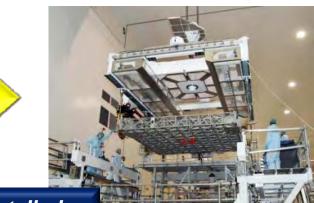
MEPSI on STS-113



ELC

(Express Logistics Carrier)Reusable external platforms on the ISS

- 8 experiment locations
- 500 pounds, > 500 watts
- High and Low rate data
- 24 month design-to-flight ready



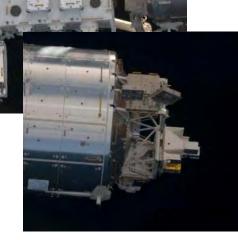
ELC 1 & 2 Installed During STS-129 mission 16 Nov 2009

JEM-EF, COL-EPF

(Japanese Experiment Module External Facility, Columbus External Payload Facility)

- International Partner external ISS sites
- 4 JEM-EF, 2 COL-EPF available to US
- Similar to ELC; JEM-EF higher mass
- 30-36 month design-to-flight ready

Both Operational





ISS - Pressurized

- Internal experiment volume
- Express Rack, power and data
- Human in the loop testing
- 18-24 month design-to-flight ready

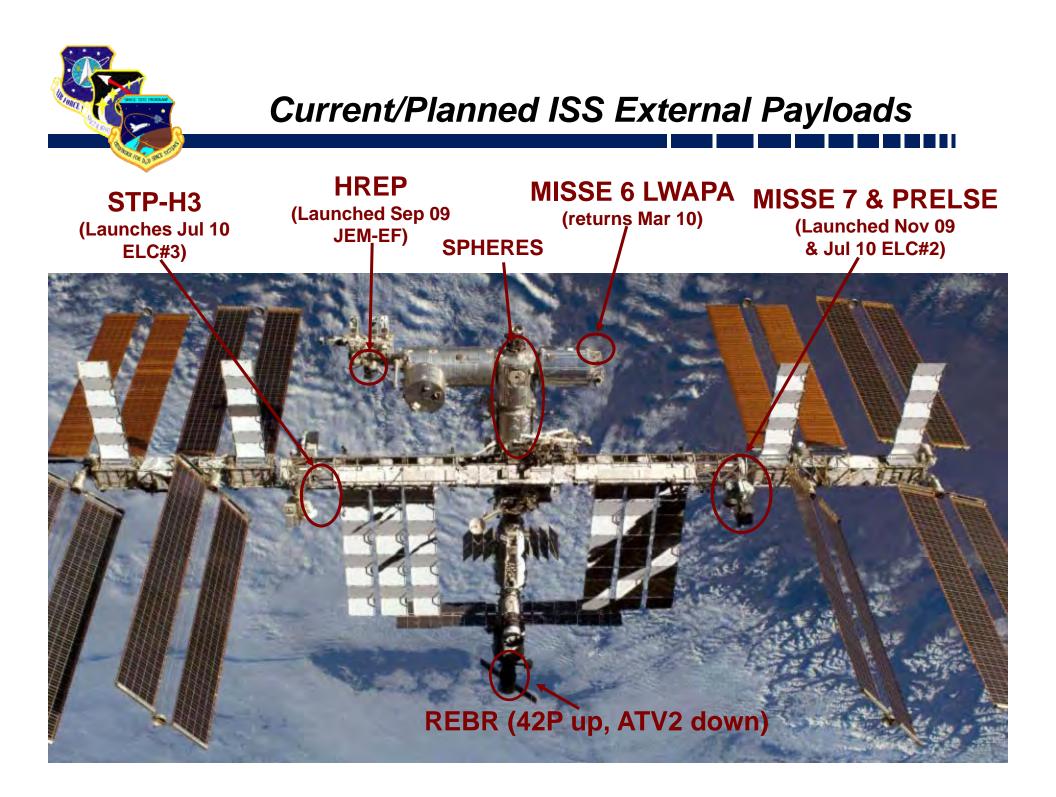


CRS

(Commercial Resupply Services)
SpaceX and Orbital Sciences Corp
Pressurized and unpressurized lift
Sample return via SpaceX Dragon
Exploring small sat deployment options



Expected to be Operational 2011





NASA International Partner Vehicles

- Japanese H-IIB Transfer Vehicle
 - Pressurized and Unpressurized
 - JEM-EF & ELC payload lift
 - No return
- European Automated Transfer Vehicle
 - Pressurized payload lift
 - No return
- Russian Progress
 - Pressurized payload lift
 - No return
- Russian Soyuz
 - Primarily crew rotation
 - Very limited payload lift
 - Very limited sample return



All Operational



• PPODs

STP Rideshare

Standardization

Reduces:

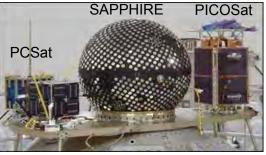
Risk

Complexity

Cost

- Why Rideshare?
 - ESPA Standard Service
 - Standard Interface Vehicle
 - M-IV Multi-Payload Adapter
 - Dedicated launches too expensive
 - Partnerships share the financial burden
- Innovation key to success
- DoD STP leads USG Rideshare Working Group
 - Explores USG, commercial & foreign launch opportunities
 - Builds partnership missions
 - Payloads include STP experiments & USG Auxiliary Payloads (APL)





Kodiak Star spacecraft suite



EELV Standard Service

- Expect one ESPA ring to fly one/year starting 2012
- Multiple ESPA class SC opportunities a year
- All APLs through STP
- Various Orbits
- Cost TBD





SECRETARY OF THE AIR FORCE WASHINGTON

MEMORANDUM FOR AFSPC/CC

February 13, 2008

SUBJECT: EELV Secondary Payload Adapter (ESPA) Policy

The Air Force has many Evolved Expendable Launch Vehicle (EELV) missions programmed across the FYDP with anticipated excess weight margin. We should leverage this excess capacity by maximizing our use of the EELV Secondary Payload Adapter (ESPA), which was successfully demonstrated in March 2007 on STP-1. As such, it is my policy to make ESPA-hosted satellite launches a routine operation starting NLT FY12.

I would like Air Force Space Command (AFSPC) to develop an ESPA utilization plan and implementation guidance in time to support the FY10 POM. AFSPC should also continue near-term efforts to make the ESPA available as a low-cost, highly reliable, standardized service for small payloads when technically feasible and consistent with overall mission assurance.

This policy is an important milestone in our efforts to provide routine and affordable access to space for scientific, research, development, and Operationally Responsive Space (ORS) missions. I look forward to your continued support in this endeavor.

cc: SAF/US AFPEO/SP AF/A3/5 AF/A8 AFRL/CC



Summary

Rapid Development & Acquisition

- Small Launch Vehicles
- Small Spacecraft
- Satellite C2 Systems
- Providers of Access to Space
 - RDT&E Launch Operations
 - RDT&E Satellite Operations
- Space Developmental Test and Evaluation
 - SDTD missions
 - Other missions (as assigned)
- End-To-End Space Mission Management
 - Mission Design
 - Systems Engineering
 - Payload Integration
 - Mission Assurance
 - Launch Operations
 - Satellite Operations
 - Transition to Operations
- Development of Space Professionals
 - Military
 - Gov't Civilians
 - OtherS



"Masters of the Calculated Risk"





QUESTIONS?

