

Entry Systems Modeling and Ground Testing: Enabling Flight Performance and Risk Reduction

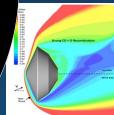


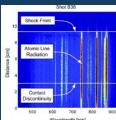
Contact: Michael Barnhardt (Michael.D.Barnhardt@nasa.gov), Aaron Brandis (Aaron.M.Brandis@nasa.gov)

Science Impact: Entry, Descent, and Landing (EDL) comprises a relatively small portion of a mission's timeline, however, it is typically among the largest risks. Flying through a body's atmosphere reliably and accurately – from orbit to ground or via aerocapture – is a critical step toward successful in situ exploration.

TPS Materials Modeling

Advanced material response models for HEEET and other next-gen woven composites; Arc jet testing for TPS performance qualification; Lab capabilities to characterize material properties Guidance, Navigation, and Control Guidance, Navigation, and Control models enable precision landing and aerocapture for scientific payloads







Aerodynamic Modeling Free-flight CFD for vehicle dynamics and aerocapture; Turbulent heating models; Parachute modeling; Wind tunnels to characterize aerodynamic performance and heating environments Shock Layer Kinetics and Radiation High-fidelity aerothermal simulation for Mars, Gas Giants, Venus, and sample return to Earth; Electric Arc Shock Tube facility measures flight-similar radiative emission



Entry Systems Modeling and Test Capabilities Applied to Planetary Science Missions



Entry, Descent, and Landing (EDL) modeling and testing are critical components of a mission's lifecycle, from concept through execution.

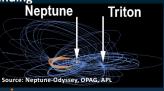
Elements of Spacecraft Flight:

ARRIVAL: Trajectory Analysis

(POST2, Genesis)

- Advanced guidance and control algorithms for precision aerocapture and landing
- Launch-to-land guidance

Multi-destination mission concepts



ENTRY: Aerothermodynamics

(Shock tube, Ballistic range, Hypersonic wind tunnels, NEQAIR. HARA)

Thermal Protection System (TPS) Modeling

(Icarus, PATO, PuMA, Hydra)

- Shock layer radiation for all destinations
- Aeroheating environment test & simulation
- Thermal protection system sizing
- Aeroshell reliability, reduced TPS margins / mission risk







Arc Jets

Parachute

simulation







Planetary Science Mission Infusion:

Small Bodies / Sample Return

Advanced models are necessary to meet stringent reliability requirements:

- Extreme aeroheating environments
- High-fidelity woven TPS modeling
- **Quantify TPS failure & reliability**
- Simulation & testing of entry vehicle dynamics

Outer Planets

New EDL challenges:

- Unique H₂-He-CH₄ aerothermochemistry
- Aerodynamics, guidance, and control for aerocapture at Ice Giants
- High-fidelity woven TPS modeling

Mars Exploration

Expanding the performance envelope:

- Reduce TPS mass margins to accommodate greater scientific payloads
- Aero, guidance, and control for precision landing to higher latitude, elevation

Venus Exploration

Extreme conditions:

- High-speed aerothermodynamics for dense CO_2/N_2 atmosphere
- High-fidelity woven TPS modeling
- Simulation & testing of entry vehicle dvnamics
- Simulation of parachute & separation dynamics

Electric Arc Shock Tube

DESCENT: Aerodynamics (Wind tunnels, Ballistic range, US3D, LAVA) Entry vehicle dynamics test & simulation

- Parachute inflation & descent dynamics
- Ground-to-flight traceability

